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## 1 1D unsteady convection

### 1.1 Propagation of a cosine profile

$$
\begin{aligned}
& \begin{cases}u_{t}+a u_{x}=0 & x \in(0,1), t \in(0,0.6] \\
u(x, 0)=u_{0}(x) & x \in(0,1) \\
u(0, t)=0 & t \in(0,0.6]\end{cases} \\
& u_{0}(x)= \begin{cases}\frac{1}{2}\left(1+\cos \left(\pi\left(x-x_{0}\right) / \sigma\right)\right) & \text { if }\left|x-x_{0}\right| \leq \sigma, \\
0 & \text { otherwise }\end{cases} \\
& a=1, x_{0}=0.2, \sigma=0.12, \Delta x=2 \cdot 10^{-2}
\end{aligned}
$$

In all the following cases, linear finite elements of Galerkin spatial discretization is used, while different time-integration methods are tested and compared.

### 1.1.1 Explicit methods

1. Second-order Lax-Wendroff with consistent mass matrix (TG2)
2. Second-order Lax-Wendroff with lumped mass matrix (LW-FD)
3. Third-order Taylor-Galerkin (TG3)

## Explicit methods results:

1. The results obtained using the different explicit methods at various values of the Courant number, $C$, are shown in Figure 1.
2. For Lax-Wendroff with consistent mass matrix (TG2), it is observed that the method is unstable at $C=0.75$, which agrees with the theory, as the stability limit for this method is $C<\frac{1}{3}$.
3. To increase the stability limit for (TG2), a lumped mass matrix is used, but this comes at the cost of accuracy. It is observed that the Lax-Wendroff method with lumped mass matrix (LW-DF) has lower phase accuracy when compared to the other two methods.
4. As for the third-order Taylor-Galerkin method (TG3), it is observed that the phase accuracy is better, and it is uniform over the entire stable interval $0<C<1$.


Figure 1: Propagation of a cosine profile using explicit methods, (Top): TG2, (Middle): LWFD, (Bottom): TG3

### 1.1.2 Implicit methods

1. Second-order Crank-Nicholson with consistent mass matrix (CN-FE)
2. Second-order Crank-Nicholson with lumped mass matrix (CN-FD)
3. Implicit fourth-order Taylor-Galerkin with consistent mass matrix (TG4)

## Implicit methods results:

1. The results obtained using the different implicit methods at various values of the Courant number, $C$, are shown in Figure 2.
2. Crank-Nicholson method is unconditionally stable. However, lower accuracy is observed at higher values of $C$.
3. Again, using lumped mass matrix leads to degradation of the phase accuracy as seen for Crank-Nicholson with lumped mass matrix (CN-FD).
4. As for the implicit fourth-order Taylor-Galerkin method (TG4), it is showing excellent phase accuracy at all values of $C$.










Figure 2: Propagation of a cosine profile using implicit methods, (Top): CN-FE, (Middle): CN-FD, (Bottom): TG4

### 1.2 Propagation of a steep front

$$
\begin{aligned}
& \begin{cases}u_{t}+a u_{x}=0 & x \in(0,1), t \in(0,0.6] \\
u(x, 0)=u_{0}(x) & x \in(0,1) \\
u(0, t)=1 & t \in(0,0.6]\end{cases} \\
& u_{0}(x)= \begin{cases}1 & \text { if } x \leq 0.2 \\
0 & \text { otherwise }\end{cases} \\
& a=1, \Delta x=2 \cdot 10^{-2}, \Delta t=1.5 \cdot 10^{-2}
\end{aligned}
$$

Under those conditions, the Courant-Friedrichs-Lewy (CFL) condition is:

$$
C=\frac{a \Delta t}{\Delta x}=\frac{1 * 0.015}{0.02}=0.75
$$

### 1.2.1 Crank-Nicholson with linear finite elements for Galerkin formulation

Employing Crank-Nicholson time integration with linear finite elements for the Galerkin spatial discretization, the obtained solution is shown in Figure 3. It is noted that even though the method is unconditionally stable, spurious oscillations appear due to the lack of artificial diffusion, which results in a non-accurate solution.


Figure 3: Propagation of steep front using the Crank-Nicholson scheme with the Galerkin method.

### 1.2.2 Crank-Nicholson with linear finite elements for least-squares formulation

Considering the linear convection equation with source term, the Crank-Nicholson time discretization is given by:

$$
\begin{equation*}
\frac{\Delta u}{\Delta t}+\frac{1}{2}(\boldsymbol{a} \cdot \boldsymbol{\nabla}) \Delta u=\frac{1}{2} s^{n+1}+\frac{1}{2} s^{n}-\boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n} \tag{1}
\end{equation*}
$$

which can be viewed as:

$$
\begin{equation*}
\mathcal{L}(\Delta u)-f=0 \tag{2}
\end{equation*}
$$

where $\mathcal{L}=\frac{1}{\Delta t}+\frac{1}{2} \boldsymbol{a} \cdot \boldsymbol{\nabla}$, and $f=\frac{1}{2} s^{n+1}+\frac{1}{2} s^{n}-\boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}$
the least-square weak formulation is written as:

$$
\begin{equation*}
(\mathcal{L}(w), \mathcal{L}(\Delta u)-f)_{\Omega}=0 \tag{3}
\end{equation*}
$$

which yields,

$$
\begin{equation*}
\left(\frac{w}{\Delta t}+\frac{1}{2} \boldsymbol{a} \cdot \boldsymbol{\nabla} w, \frac{\Delta u}{\Delta t}+\frac{1}{2} \boldsymbol{a} \cdot \boldsymbol{\nabla} \Delta u\right)_{\Omega}=\left(\frac{w}{\Delta t}+\frac{1}{2} \boldsymbol{a} \cdot \boldsymbol{\nabla} w, \frac{1}{2} s^{n+1}+\frac{1}{2} s^{n}-\boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega} \tag{4}
\end{equation*}
$$

further manipulation,

$$
\begin{align*}
\frac{1}{\Delta t^{2}}(w, \Delta u)_{\Omega} & +\frac{1}{2 \Delta t}(w, \boldsymbol{a} \cdot \boldsymbol{\nabla} \Delta u)_{\Omega}+\frac{1}{2 \Delta t}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \Delta u)_{\Omega}+\frac{1}{4}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} \Delta u)_{\Omega} \\
= & \frac{1}{2 \Delta t}\left(w, s^{n+1}\right)_{\Omega}+\frac{1}{2 \Delta t}\left(w, s^{n}\right)_{\Omega}-\frac{1}{\Delta t}\left(w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega}+\frac{1}{4}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, s^{n+1}\right)_{\Omega}  \tag{5}\\
& +\frac{1}{4}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, s^{n}\right)_{\Omega}-\frac{1}{2}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega}
\end{align*}
$$

assuming that the source term is not time-dependent $\left(s^{n+1}=s^{n}=s\right)$ yields,

$$
\begin{align*}
\frac{1}{\Delta t^{2}}(w, \Delta u)_{\Omega} & +\frac{1}{2 \Delta t}(w, \boldsymbol{a} \cdot \boldsymbol{\nabla} \Delta u)_{\Omega}+\frac{1}{2 \Delta t}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \Delta u)_{\Omega}+\frac{1}{4}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} \Delta u)_{\Omega}  \tag{6}\\
= & -\frac{1}{\Delta t}\left(w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega}-\frac{1}{2}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega}+\frac{1}{\Delta t}(w, s)_{\Omega}+\frac{1}{2}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, s)_{\Omega}
\end{align*}
$$

integrating by parts the second term on LHS and the first term on RHS yields:

$$
\begin{align*}
\frac{1}{\Delta t^{2}}(w, \Delta u)_{\Omega}- & \frac{1}{2 \Delta t}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \Delta u)_{\Omega}+\frac{1}{2 \Delta t}((\boldsymbol{a} \cdot \boldsymbol{n}) w, \Delta u)_{\Gamma} \\
& +\frac{1}{2 \Delta t}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \Delta u)_{\Omega}+\frac{1}{4}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} \Delta u)_{\Omega} \\
= & \frac{1}{\Delta t}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n}\right)_{\Omega}-\frac{1}{\Delta t}\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n}\right)_{\Gamma}-\frac{1}{2}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega}  \tag{7}\\
& +\frac{1}{\Delta t}(w, s)_{\Omega}+\frac{1}{2}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, s)_{\Omega}
\end{align*}
$$

given that the boundary $\Gamma=\Gamma^{i n} \cup \Gamma^{\text {out }}$, and recalling that for hyperbolic problems the boundary conditions are applied only to inflow boundaries. Furthermore, the inflow boundary could be a Neumann part and a Dirchlet part, i.e. $\Gamma^{i n}=\Gamma_{N}^{i n} \cup \Gamma_{D}^{i n}$. Since $w=0$ on the Dirchlet boundary, therefore the weak form is written as:

$$
\begin{align*}
\frac{1}{\Delta t^{2}}(w, \Delta u)_{\Omega}+ & \frac{1}{2 \Delta t}((\boldsymbol{a} \cdot \boldsymbol{n}) w, \Delta u)_{\Gamma^{\text {out }}}+\frac{1}{2 \Delta t}((\boldsymbol{a} \cdot \boldsymbol{n}) w, \Delta u)_{\Gamma_{N}^{i n}}+\frac{1}{4}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} \Delta u)_{\Omega} \\
= & \frac{1}{\Delta t}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n}\right)_{\Omega}-\frac{1}{\Delta t}\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n}\right)_{\Gamma^{\text {out }}}-\frac{1}{\Delta t}\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n}\right)_{\Gamma_{N}^{i n}}-\frac{1}{2}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega} \\
& +\frac{1}{\Delta t}(w, s)_{\Omega}+\frac{1}{2}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, s)_{\Omega} \tag{8}
\end{align*}
$$

particularizing it to the problem at hand (no Neumann boundary) yields,

$$
\begin{align*}
\frac{1}{\Delta t^{2}}(w, \Delta u)_{\Omega}+ & \frac{1}{2 \Delta t}((\boldsymbol{a} \cdot \boldsymbol{n}) w, \Delta u)_{\Gamma^{\text {out }}}+\frac{1}{4}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} \Delta u)_{\Omega} \\
= & \frac{1}{\Delta t}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n}\right)_{\Omega}-\frac{1}{\Delta t}\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n}\right)_{\Gamma^{\text {out }}}-\frac{1}{2}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega}  \tag{9}\\
& +\frac{1}{\Delta t}(w, s)_{\Omega}+\frac{1}{2}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, s)_{\Omega}
\end{align*}
$$

neglecting the outflow boundary term, because the wave front doesn't reach it, yields,

$$
\begin{align*}
\frac{1}{\Delta t^{2}}(w, \Delta u)_{\Omega}+\frac{1}{4}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} \Delta u)_{\Omega}= & \frac{1}{\Delta t}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n}\right)_{\Omega}-\frac{1}{2}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega}  \tag{10}\\
& +\frac{1}{\Delta t}(w, s)_{\Omega}+\frac{1}{2}(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, s)_{\Omega}
\end{align*}
$$

which is written in discrete form as:

$$
\begin{equation*}
\frac{1}{\Delta t^{2}} \boldsymbol{M} \boldsymbol{\Delta} \boldsymbol{U}+\frac{1}{4} \boldsymbol{K} \boldsymbol{\Delta} \boldsymbol{U}=\frac{1}{\Delta t} \boldsymbol{C} \boldsymbol{U}^{n}-\frac{1}{2} \boldsymbol{K} \boldsymbol{U}^{n}+\frac{1}{\Delta t} \boldsymbol{v}_{\boldsymbol{1}}+\frac{1}{2} \boldsymbol{v}_{\boldsymbol{2}} \tag{11}
\end{equation*}
$$

where the appearing vectors are defined as:

$$
\begin{align*}
\boldsymbol{M} \boldsymbol{U} & =(w, u)_{\Omega}, \\
\boldsymbol{C} \boldsymbol{U} & =(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u)_{\Omega}, \\
\boldsymbol{K} \boldsymbol{U} & =(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u)_{\Omega},  \tag{12}\\
\boldsymbol{v}_{\mathbf{1}} & =(w, s)_{\Omega} \\
\boldsymbol{v}_{\mathbf{2}} & =(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, s)_{\Omega},
\end{align*}
$$

equation (11) is further simplified to:

$$
\begin{equation*}
\left[\boldsymbol{M}+\frac{\Delta t^{2}}{4} \boldsymbol{K}\right] \boldsymbol{\Delta} \boldsymbol{U}=\left[\Delta t \boldsymbol{C}-\frac{\Delta t^{2}}{2} \boldsymbol{K}\right] \boldsymbol{U}^{n}+\left[\Delta t \boldsymbol{v}_{\mathbf{1}}+\frac{\Delta t^{2}}{2} \boldsymbol{v}_{\mathbf{2}}\right] \tag{13}
\end{equation*}
$$

and for ease of implementation in Matlab, it is written as:

$$
\boldsymbol{A} \boldsymbol{\Delta} \boldsymbol{U}=\boldsymbol{B} \boldsymbol{U}^{n}+\boldsymbol{f}
$$

and the corresponding part of the Matlab function (system_CN_LS.m) is:

```
1 % Matrices assembly
2 A(isp,isp) = A(isp,isp) + w_ig*(N'*N + dt_2^2*(a*Nx)'* (a*Nx));
3 B(isp,isp) = B(isp,isp) + w_ig*(dt*(a*Nx)'*N - (dt^2/2)*(a*Nx)'* (a*Nx));
4 f(isp) = f(isp) + w_ig*(dt*(N')*SourceTerm(x) + ...
    (dt^2/2)* (a*Nx)'*SourceTerm(x));
```

Now, using the least-square formulation (linear finite elements) with Crank-Nicholson time integration scheme gives the solution shown in Figure 4b. It is observed that Crank-Nicholson with least-squares succeeded in removing the spurious oscillation induced by the Galerkin formulation over the whole computational domain.

There still exist some residual oscillations at the front because Crank-Nicholson scheme is not a monotone scheme [1]. These could be removed by adding more artificial viscosity that locally (only) at the front.


Figure 4: Propagation of steep front using the Crank-Nicholson scheme with Galerkin formulation (left) and with the least-squares formulation (right).

### 1.2.3 Lax-Wendroff with linear finite elements for Galerkin formulation

Solving the problem using the second-order Lax-Wendroff scheme with consistent mass matrix (TG2) yields unstable solution as seen in Figure 5. The reason is that Lax-Wendroff is conditionally stable, where the stability condition is $C^{2}<\frac{1}{3}$, i.e $|C|<0.5774$.


Figure 5: Propagation of steep front using the second-order Lax-Wendroff scheme with consistent mass matrix (TG2) and with the Galerkin method.

A way to increase the stability limit to $|C|<1$ is to use lumped mass matrix. The results using the second-order Lax-Wendroff scheme with lumped mass matrix (LW-FD) yields stable solution as seen in Figure 6.


Figure 6: Propagation of steep front using the second-order Lax-Wendroff scheme with lumped mass matrix (LW-FD) and with the Galerkin method.

### 1.2.4 Two-step Lax-Wendroff with linear finite elements for Galerkin formulation

The time discretization of the second-order two-step Lax-Wendroff scheme, also called Richtmyer scheme, is given by:

$$
\begin{align*}
& u^{n+\frac{1}{2}}=u^{n}+\frac{\Delta t}{2} u_{t}^{n},  \tag{14}\\
& u^{n+1}=u^{n}+\Delta t u_{t}^{n+\frac{1}{2}}
\end{align*}
$$

recalling the linear advection equation:

$$
\begin{equation*}
u_{t}^{n}=s^{n}-\boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n} \tag{15}
\end{equation*}
$$

substituting (15) into (14) yields:

$$
\begin{align*}
u^{n+\frac{1}{2}} & =u^{n}+\frac{\Delta t}{2} s^{n}-\frac{\Delta t}{2} \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}  \tag{16a}\\
u^{n+1} & =u^{n}+\Delta t s^{n+\frac{1}{2}}-\Delta t \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n+\frac{1}{2}} \tag{16b}
\end{align*}
$$

assuming that the source term is not time dependent, then the Galerkin weak form is given by:

$$
\begin{align*}
& \left(w, u^{n+\frac{1}{2}}\right)_{\Omega}=\left(w, u^{n}\right)_{\Omega}+\frac{\Delta t}{2}(w, s)_{\Omega}-\frac{\Delta t}{2}\left(w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n}\right)_{\Omega}  \tag{17a}\\
& \left(w, u^{n+1}\right)_{\Omega}=\left(w, u^{n}\right)_{\Omega}+\Delta t(w, s)_{\Omega}-\Delta t\left(w, \boldsymbol{a} \cdot \boldsymbol{\nabla} u^{n+\frac{1}{2}}\right)_{\Omega} \tag{17b}
\end{align*}
$$

integration by parts and applying Dirchlet boundary condition at the inlet yields:

$$
\begin{align*}
\left(w, u^{n+\frac{1}{2}}\right)_{\Omega}=\left(w, u^{n}\right)_{\Omega}+\frac{\Delta t}{2}(w, s)_{\Omega}+\frac{\Delta t}{2}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n}\right)_{\Omega} & -\frac{\Delta t}{2}\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n}\right)_{\Gamma^{\text {out }}} \\
& -\frac{\Delta t}{2}\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n}\right)_{\Gamma_{N}^{i n}}  \tag{18a}\\
\left(w, u^{n+1}\right)_{\Omega}=\left(w, u^{n}\right)_{\Omega}+\Delta t(w, s)_{\Omega}+\Delta t\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n+\frac{1}{2}}\right)_{\Omega} & -\Delta t\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n+\frac{1}{2}}\right)_{\Gamma^{\text {out }}} \\
& -\Delta t\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n+\frac{1}{2}}\right)_{\Gamma_{N}^{i n}} \tag{18b}
\end{align*}
$$

particularizing it to the problem at hand (no Neumann boundary) yields:

$$
\begin{align*}
& \left(w, u^{n+\frac{1}{2}}\right)_{\Omega}=\left(w, u^{n}\right)_{\Omega}+\frac{\Delta t}{2}(w, s)_{\Omega}+\frac{\Delta t}{2}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n}\right)_{\Omega}-\frac{\Delta t}{2}\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n}\right)_{\Gamma^{\circ o u t}}  \tag{19a}\\
& \left(w, u^{n+1}\right)_{\Omega}=\left(w, u^{n}\right)_{\Omega}+\Delta t(w, s)_{\Omega}+\Delta t\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n+\frac{1}{2}}\right)_{\Omega}-\Delta t\left((\boldsymbol{a} \cdot \boldsymbol{n}) w, u^{n+\frac{1}{2}}\right)_{\Gamma^{o u t}} \tag{19b}
\end{align*}
$$

neglecting the outflow boundary term, because the wave front doesn't reach it, yields, particularizing it to the problem at hand (no Neumann boundary) yields:

$$
\begin{align*}
& \left(w, u^{n+\frac{1}{2}}\right)_{\Omega}=\left(w, u^{n}\right)_{\Omega}+\frac{\Delta t}{2}(w, s)_{\Omega}+\frac{\Delta t}{2}\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n}\right)_{\Omega}  \tag{20a}\\
& \left(w, u^{n+1}\right)_{\Omega}=\left(w, u^{n}\right)_{\Omega}+\Delta t(w, s)_{\Omega}+\Delta t\left(\boldsymbol{a} \cdot \boldsymbol{\nabla} w, u^{n+\frac{1}{2}}\right)_{\Omega} \tag{20b}
\end{align*}
$$

recalling the definitions given by (12), the discrete form is written as:

$$
\begin{align*}
& \boldsymbol{M} \boldsymbol{U}^{n+\frac{1}{2}}=\boldsymbol{M} \boldsymbol{U}^{n}+\frac{\Delta t}{2} \boldsymbol{v}_{\mathbf{1}}+\frac{\Delta t}{2} \boldsymbol{C} \boldsymbol{U}^{n}  \tag{21a}\\
& \boldsymbol{M} \boldsymbol{U}^{n+1}=\boldsymbol{M} \boldsymbol{U}^{n}+\Delta t \boldsymbol{v}_{\mathbf{1}}+\Delta t \boldsymbol{C} \boldsymbol{U}^{n+\frac{1}{2}} \tag{21b}
\end{align*}
$$

further simplification yields:

$$
\begin{align*}
& \boldsymbol{M}\left(\boldsymbol{U}^{n+\frac{1}{2}}-\boldsymbol{U}^{n}\right)=\frac{\Delta t}{2} \boldsymbol{C} \boldsymbol{U}^{n}+\frac{\Delta t}{2} \boldsymbol{v}_{\boldsymbol{1}}  \tag{22a}\\
& \boldsymbol{M}\left(\boldsymbol{U}^{n+1}-\boldsymbol{U}^{n}\right)=\Delta t \boldsymbol{C} \boldsymbol{U}^{n+\frac{1}{2}}+\Delta t \boldsymbol{v}_{\boldsymbol{1}} \tag{22b}
\end{align*}
$$

and for ease of implementation in Matlab, it is written as:

$$
\begin{align*}
\boldsymbol{A}\left(\boldsymbol{U}^{n+\frac{1}{2}}-\boldsymbol{U}^{n}\right) & =\frac{1}{2} \boldsymbol{B} \boldsymbol{U}^{n}+\frac{1}{2} \boldsymbol{f},  \tag{23a}\\
\boldsymbol{A}\left(\boldsymbol{U}^{n+1}-\boldsymbol{U}^{n}\right) & =\boldsymbol{B} \boldsymbol{U}^{n+\frac{1}{2}}+\boldsymbol{f} \tag{23b}
\end{align*}
$$

and the corresponding part of the Matlab function (system_LW_2S.m) is:

```
% Matrices assembly
A(isp,isp) = A(isp,isp) + w_ig*N'*N;
B(isp,isp) = B(isp,isp) + w_ig*dt*(a*Nx)'*N;
f(isp) = f(isp) + w_ig*dt_2*(N')*SourceTerm(x);
```

and the modification made in the Main.m to account for the two-steps is as follows:

```
% SOLUTION AT EACH TIME STEP
if meth == 7 || meth == 8 % two-step Lax-Wendroff methods
    for n = 1:nstep
            % step 1
            btot_1 = [0.5*B*u(:,n)+0.5*f; bccd];
            aux_1 = U\(L\btot_1);
            u(:,n+1) = u(:,n) + aux_1(1:numnp);
            % step 2
            btot_2 = [B*u(:,n+1)+f; bccd];
            aux_2 = U\(L\btot_2);
            u(:,n+1) = u(:,n) + aux_2(1:numnp);
        end
else
```

Using the two-step Lax-Wendroff method with consistent mass matrix yields unstable solution as seen in Figure 7a. This is expected because the stability limit is exceeded, where the Courant number is $C=0.75$ while the limit is $|C|<0.5774$. The solution for the same method but with lumped mass matrix is shown in Figure 7b. It is observed that the stability is better, but the solution is still not satisfactory. In fact, the stability limit of the two-step Lax-Wendroff method applied to the steep-front problem was found to be extremely reduced, see Figure 8 where $C=0.03,0.0003$ and yet the solution is not improving.


Figure 7: Propagation of steep front using the two-step Lax-Wendroff time integration scheme with consistent mass matrix (left) and with lumped mass matrix (right) combined with Galerkin spatial formulation.


Figure 8: Propagation of steep front using the two-step Lax-Wendroff time integration scheme with consistent mass matrix combined with Galerkin spatial formulation at very small values of Courant number.

## 2 1D unsteady convection-diffusion

$$
\left.\begin{array}{l} 
\begin{cases}u_{t}+a u_{x}-\nu u_{x x}=0 & x \in(0,1), t \in(0,0.6] \\
u(x, 0)=u_{0}(x) & x \in(0,1) \\
u(0, t)=u(1, t)=0 & t \in(0,0.6]\end{cases} \\
u_{0}(x)=\frac{5}{7} \exp \left\{-\left(x-x_{0}\right)^{2} / L^{2}\right\}
\end{array}\right\} \begin{aligned}
& a=1, x_{0}=2 / 15, L=7 \sqrt{2} / 300, \Delta x=1 / 150 \\
& \nu=\left[3.3 \cdot 10^{-3}, 6.7 \cdot 10^{-4}, 3.3 \cdot 10^{-5}\right]^{T}
\end{aligned}
$$

### 2.1 Galerkin and Crank-Nicholson

The solution is shown in Figure 9 at Courant number $C=1$. It is noticed that the second-order Crank-Nicholson scheme gives good results at low and moderate values of Peclet number as in Figures 9a and 9b, but shows significant phase error when the Peclet number is increased, i.e. convection dominated cases as in Figure 9c, and it becomes more worse as the Courant number exceeds 1 as in Figure 9d. The reason is that linear finite elements in standard Galerkin formulation do not work well with the second-order Crank-Nicholson scheme in convection dominated problems.


Figure 9: Gaussian hill: standard Galerkin combined with second-order Crank-Nicholson.

### 2.2 Galerkin and $\mathrm{R}_{2,2}$

The solution is shown in Figure 10 at higher Courant number $C=3$. It is noticed that the higher-order schemes in time provide a gain in accuracy even for convection dominated cases (high Peclet number). It is also noted that the phase accuracy is reduced as the Courant number increases as seen in Figure 10d.


Figure 10: Gaussian hill: standard Galerkin combined with fourth-order $\mathrm{R}_{2,2}$.

### 2.3 Galerkin and $\mathrm{R}_{3,3}$

The solution is shown in Figure 11 at even higher Courant number $C=4$. Again, It is noticed that the higher-order schemes in time provide a gain in accuracy even for convection dominated cases (high Peclet number). It is also noted that the phase accuracy is not significantly affected as the Courant number increases as seen in Figure 11d.


Figure 11: Gaussian hill: standard Galerkin combined with $\mathrm{R}_{3,3}$.

### 2.4 Galerkin and Adams-Bashforth

Adams-Bashforth is a second-order accurate explicit time integration method which is given by:

$$
\begin{equation*}
u^{n+1}=u^{n}+\frac{\Delta t}{2}\left(3 u_{t}^{n}-u_{t}^{n-1}\right) \tag{24}
\end{equation*}
$$

where $u_{t}=s-a u_{x}+\nu u_{x x}$. By substituting $u_{t}$ into the previous equation, the scheme is written for the unsteady convection-diffusion problem as:

$$
\begin{equation*}
\frac{2}{\Delta t} \Delta u=3 s^{n}-3 a u_{x}^{n}+3 \nu u_{x x}^{n}-s^{n-1}+a u_{x}^{n-1}-\nu u_{x x}^{n-1} \tag{25}
\end{equation*}
$$

Assuming that the source term is not time dependent yields:

$$
\begin{equation*}
\frac{2}{\Delta t} \Delta u=2 s-3 a u_{x}^{n}+3 \nu u_{x x}^{n}+a u_{x}^{n-1}-\nu u_{x x}^{n-1} \tag{26}
\end{equation*}
$$

The Galerkin weak form is then written as:

$$
\begin{equation*}
\frac{2}{\Delta t}(w, \Delta u)_{\Omega}=2(w, s)_{\Omega}-3\left(w, a u_{x}^{n}\right)_{\Omega}+3\left(w, \nu u_{x x}^{n}\right)_{\Omega}+\left(w, a u_{x}^{n-1}\right)_{\Omega}-\left(w, \nu u_{x x}^{n-1}\right)_{\Omega} \tag{27}
\end{equation*}
$$

Integrating by parts the diffusion terms and removing the boundary terms because only Dirchlet boundary conditions exists, the weak form reads:

$$
\begin{equation*}
\frac{2}{\Delta t}(w, \Delta u)_{\Omega}=2(w, s)_{\Omega}-3\left(w, a u_{x}^{n}\right)_{\Omega}-3\left(w_{x}, \nu u_{x}^{n}\right)_{\Omega}+\left(w, a u_{x}^{n-1}\right)_{\Omega}+\left(w_{x}, \nu u_{x}^{n-1}\right)_{\Omega} \tag{28}
\end{equation*}
$$

The discrete form is written as:

$$
\begin{equation*}
\boldsymbol{M} \boldsymbol{\Delta} \boldsymbol{U}=\Delta t\left(\boldsymbol{M} \boldsymbol{s}-\frac{3 a}{2} \boldsymbol{C} \boldsymbol{U}^{n}-\frac{3 \nu}{2} \boldsymbol{K} \boldsymbol{U}^{n}+\frac{a}{2} \boldsymbol{C} \boldsymbol{U}^{n-1}+\frac{\nu}{2} \boldsymbol{K} \boldsymbol{U}^{n-1}\right) \tag{29}
\end{equation*}
$$

where $M_{i j}=\left(N_{i}, N_{j}\right)_{\Omega}, C_{i j}=\left(N_{i}, N_{x j}\right)_{\Omega}$, and $K_{i j}=\left(N_{x i}, N_{x j}\right)_{\Omega}$. Here, the source term is also approximated following the provided code.

It is seen from (29) that Adams-Bashforth is not a self-starting method For this, a selfstarting method is needed in the first time step, the explicit forward Euler method is used with very small time step value. The discrete form of forward Euler scheme for convection-diffusion equation can be shown as:

$$
\begin{equation*}
\boldsymbol{M} \boldsymbol{\Delta} \boldsymbol{U}=\Delta t\left(\boldsymbol{M} \boldsymbol{s}-a \boldsymbol{C} \boldsymbol{U}^{n}-\nu \boldsymbol{K} \boldsymbol{U}^{n}\right) \tag{30}
\end{equation*}
$$

and the modification made in the Main.m to account for the two-steps is as follows:

```
if d_temp == 3 % Adams-Bashforth
    M_reduced = M (2:end-1, 2:end-1);
    Sol = c;
    for i=1:nstep
        if i == 1 % use forward Euler for 1st step
            rhs = dt*(M*f - a*C*C - nu*K*C);
            dc = M_reduced\rhs(2:end-1);
            c(2:end-1) = c(2:end-1) + dc;
            Sol = [SOl c];
            else % use Adams-Bashforth for the rest of steps
                rhs = dt*(M*f - 1.5*a*C*Sol(:,end) - 1.5*nu*K*Sol(:,end) + ...
                    0.5*a*C*Sol(:, end-1) + 0.5*nu*K*Sol(:, end-1));
            dc = M_reduced\rhs(2:end-1);
                c(2:end-1) = c(2:end-1) + dc;
                Sol = [Sol c];
        end
    end
else
```

By implementing this method and analysing it, it is noticed that based on the value of Peclet number, this method gives accurate results only at low values of the Courant number, and surprisingly tends to be more unstable for diffusion-dominated case (low values of Pe ).

Diffusion-dominated ( $P e=0.33$ ): the stability is only achieved at very low values of the Courant number around $C=0.037$ as in Figure 12a. Slightly increasing the value of the Courant number to $C=0.06$ yields unstable results in Figure 12b.

Balanced convection-diffusion $(P e=1)$ : the stability is only achieved at low values of the Courant number around $C=0.1$ as in Figure 12c. Slightly increasing the value of the Courant number to $C=0.2$ yields unstable results in Figure 12d.

Convection-dominated $(P e=5)$ : Stable solution was obtained at $C=0.4$ as in Figure 12e. Slightly increasing the Courant number to $C=0.5$ leads to instabilities in the solution as shown in Figure 12f.

Highly convection-dominated $(P e=100)$ : Stable solution was obtained at $C=0.3$ as in Figure 12g. Slightly increasing the Courant number to $C=0.4$ leads to unstable solution as shown in Figure 12h.


Figure 12: Gaussian hill: standard Galerkin combined with Adams-Bashforth.

### 2.5 Time-discontinuous Galerkin formulation for the convection-diffusion equation

Following the steps in [2] done to formulate the time-discontinuous Galerkin for the convection equation, similarly, the time-discontinuous Galerkin for the convection-diffusion equation is formulated.

Considering a piecewise continuous approximation in space and discontinuous approximation in time, the time domain is partitioned in $n_{s t}$ sub-intervals, where each sub-interval is defined as $\left.I^{n}=\right] t^{n}, t^{n+1}\left[\right.$, for $n=0,1, \ldots, n_{s t}-1$.

Space-time slabs are then obtained in the form:

$$
Q^{n}=\Omega \times I^{n}
$$

For the considered space-time slab $Q^{n}$, the spatial domain $\Omega$ is subdivided into $n_{e l}$ elements, $\Omega^{e}, e=1, \ldots, n_{e l}$, giving space-time element domains

$$
Q_{e}^{n}=\Omega^{e} \times I^{n}, \quad e=1, \ldots, n_{e l}
$$

Considering the following notation

$$
u^{h}\left(t_{ \pm}^{n}\right)=\lim _{\epsilon \rightarrow 0^{+}} u^{h}\left(t^{n} \pm \epsilon\right)
$$

The weak form of the convection-diffusion equation with zero source term is written as:

$$
\begin{equation*}
\iint_{Q^{n}} w^{h}\left(u_{t}^{h}+\boldsymbol{a} \cdot \boldsymbol{\nabla} u^{h}-\boldsymbol{\nabla} \cdot\left(\nu \boldsymbol{\nabla} u^{h}\right)\right) d \Omega d t+\int_{\Omega} w^{h}\left(t_{+}^{n}\right)\left(u^{h}\left(t_{+}^{n}\right)-u^{h}\left(t_{-}^{n}\right)\right) d \Omega=0 \tag{31}
\end{equation*}
$$

with the initial condition $u^{h}\left(\boldsymbol{x}, t_{-}^{0}\right)=u_{0}(\boldsymbol{x})$.
Integrating by parts and considering only Dirchlet boundary conditions yields:

$$
\begin{equation*}
\iint_{Q^{n}} w^{h}\left(u_{t}^{h}+\boldsymbol{a} \cdot \boldsymbol{\nabla} u^{h}\right) d \Omega d t+\iint_{Q^{n}} \boldsymbol{\nabla} w^{h} \cdot \nu \boldsymbol{\nabla} u^{h} d \Omega d t+\int_{\Omega} w^{h}\left(t_{+}^{n}\right)\left(u^{h}\left(t_{+}^{n}\right)-u^{h}\left(t_{-}^{n}\right)\right) d \Omega=0 \tag{32}
\end{equation*}
$$

Using finite element approximations over a space-time slab which are piecewise polynomials in space and linear in time; that is, for $(\boldsymbol{x}, t) \in Q^{n}=\Omega \times I^{n}$,

$$
u^{h}(\boldsymbol{x}, t)=\sum_{A=1}^{n_{n p}} N_{A}(\boldsymbol{x})\left(\Theta_{1}(t) \tilde{\mathrm{u}}_{A}^{n}+\Theta_{2}(t) \mathrm{u}_{A}^{n+1}\right)
$$

where $N_{A}(\boldsymbol{x})$ is the spatial shape function at node $A ; \tilde{\mathrm{u}}_{A}^{n}$ and $\mathrm{u}_{A}^{n+1}$ are the nodal values of $u^{h}$ for node $A$ at $t_{+}^{n}$ and $t_{-}^{n+1}$, respectively. $\Theta_{1}(t)$ and $\Theta_{2}(t)$ are the time interpolation functions defined linearly as

$$
\begin{aligned}
& \Theta_{1}(t)=\frac{t^{n+1}-t}{t^{n+1}-t^{n}}=\frac{t^{n+1}-t}{\Delta t} \\
& \Theta_{2}(t)=\frac{t-t^{n}}{t^{n+1}-t^{n}}=\frac{t-t^{n}}{\Delta t}
\end{aligned}
$$

The test function $w^{h}$ for each time slab (piecewise polynomials in space and linear in time) are similarly defined, $N_{A}(\boldsymbol{x}) \Theta_{1}(t)$ and $N_{A}(\boldsymbol{x}) \Theta_{2}(t)$ for $A=1, \ldots, n_{n p}$.

With these definitions, the weak form (31) yields the following couple of equations for each node $A$ :

$$
\begin{align*}
& \sum_{B=1}^{n_{n p}}\left\{\int \int _ { Q ^ { n } } N _ { A } \Theta _ { 1 } \left[N_{B} \frac{\mathrm{u}_{B}^{n+1}-\tilde{\mathrm{u}}_{B}^{n}}{\Delta t}+\right.\right.\left.\left(\Theta_{1} \tilde{\mathrm{u}}_{B}^{n}+\Theta_{2} \mathrm{u}_{B}^{n+1}\right)(\boldsymbol{a} \cdot \boldsymbol{\nabla}) N_{B}\right] d \Omega d t  \tag{33a}\\
&\left.+\iint_{Q^{n}} \nabla N_{A} \Theta_{1} \nu\left(\Theta_{1} \tilde{\mathrm{u}}_{B}^{n}+\Theta_{2} \mathrm{u}_{B}^{n+1}\right) \nabla N_{B}\right\}=0 \\
& \sum_{B=1}^{n_{n p}}\left\{\iint_{Q^{n}} N_{A} \Theta_{2}\left[N_{B} \frac{\mathrm{u}_{B}^{n+1}-\tilde{\mathrm{u}}_{B}^{n}}{\Delta t}+\left(\Theta_{1} \tilde{\mathrm{u}}_{B}^{n}+\Theta_{2} \mathrm{u}_{B}^{n+1}\right)(\boldsymbol{a} \cdot \nabla) N_{B}\right] d \Omega d t\right. \\
&\left.+\iint_{Q^{n}} \nabla N_{A} \Theta_{2} \nu\left(\Theta_{1} \tilde{\mathrm{u}}_{B}^{n}+\Theta_{2} \mathrm{u}_{B}^{n+1}\right) \nabla N_{B}\right\}  \tag{33b}\\
&+\int_{Q^{n}} N_{A} \sum_{B=1}^{n_{n p}} N_{B}\left(\tilde{\mathrm{u}}_{B}^{n}-\mathrm{u}_{B}^{n}\right) d \Omega=0
\end{align*}
$$

Following the Matrix form for the convection equation shown in [3], The Matrix form for the convection-diffusion equation is written, where the unknowns are $\mathbf{u}^{n+1}$ and $\mathbf{u}^{n^{+}}$, as:

$$
\begin{align*}
& \left(\boldsymbol{M}+\frac{2}{3} \Delta t \boldsymbol{C}+\frac{2 \nu}{3} \Delta t \boldsymbol{K}\right) \mathbf{u}^{n+1}-\left(\boldsymbol{M}-\frac{1}{3} \Delta t \boldsymbol{C}-\frac{\nu}{3} \Delta t \boldsymbol{K}\right) \mathbf{u}^{n^{+}}=\mathbf{0}  \tag{34a}\\
& \left(\boldsymbol{M}+\frac{1}{3} \Delta t \boldsymbol{C}+\frac{\nu}{3} \Delta t \boldsymbol{K}\right) \mathbf{u}^{n+1}+\left(\boldsymbol{M}+\frac{2}{3} \Delta t \boldsymbol{C}+\frac{2 \nu}{3} \Delta t \boldsymbol{K}\right) \mathbf{u}^{n^{+}}=2 \boldsymbol{M} \mathbf{u}^{n^{-}} \tag{34b}
\end{align*}
$$

## A Developed codes for 1D unsteady convection

## A. 1 New function system_CN_LS.m

This function includes the implementation of the least-square spatial formulation combined with Crank-Nicholson time integration scheme.

```
function [A,B,f] = system_CN_LS(xnode,a)
% [A,B,f] = system_CN_LS(xnode,a)
% L.h.s (A) and r.h.s (B,f) matrices for the second-order
% implicit Crank-Nicolson scheme using the consistent mass matrix.
%
% Least-Squares spatial discretization is used
%
% xnode: nodal coordinates
% a : velocity
%
global dt
dt_2 = dt/2;
% Gauss points and weights on the reference element [-1,1]
xipg = [-1/sqrt(3) 1/sqrt(3)]';
wpg = [1 1]';
% Shape functions and its derivatives in the reference element
N_mef = [(1-xipg)/2 (1+xipg)/2];
Nxi_mef = [-1/2 1/2; -1/2 1/2];
% Total number of nodes and elements
numnp = size(xnode,2);
numel = numnp-1;
% Number of Gauss points on an element
ngaus = size(wpg,1);
% Allocate storage
A = zeros(numnp,numnp);
B = zeros(numnp, numnp);
f = zeros(numnp,1);
% MATRICES COMPUTATION
% Loop on elements
for i=1:numel
    unos = ones (ngaus,1);
    h = xnode(i+1)-xnode(i);
    xm = (xnode(i) +xnode(i+1))/2;
    weight = wpg*h/2;
    isp = [i i+1];
    % Loop on Gauss points (numerical quadrature)
    for ig=1:ngaus
        N = N_mef(ig,:);
        Nx = Nxi_mef(ig,:)*2/h;
        w_ig = weight(ig);
```

```
    x = xm + h/2*xipg(ig); % x-coordinate of the current Gauss point
    % Matrices assembly
    A(isp,isp) = A(isp,isp) + w_ig*(N'*N + dt_2^2*(a*Nx)'* (a*Nx));
    B(isp,isp) = B(isp,isp) + w_ig*(dt*(a*Nx)'*N - ...
        (dt^2/2)* (a*Nx)'* (a*Nx));
    f(isp) = f(isp) + w_ig*(dt* (N')*SourceTerm(x) + ...
        (dt^2/2)* (a*Nx)'*SourceTerm(x));
    end
    end
```


## A. 2 New function system_LW_2S.m

This function includes the implementation of the two-step Lax-Wendroff time integration scheme with Galerkin spatial formulation.

```
function [A,B,f] = system_LW_2S(xnode,a)
% [A,B,f] = system_LW_2S (xnode,a)
% L.h.s (A) and r.h.s (B,f) matrices for the two-step Lax-Wendroff scheme
% with consistent mass matrix
%
% xnode: nodal coordinates
% a : velocity
%
global dt
dt_2 = dt/2;
% Gauss points and weights on the reference element [-1,1]
xipg = [-1/sqrt(3) 1/sqrt(3)]';
wpg = [1 1]';
% Shape functions and its derivatives in the reference element
N_mef = [(1-xipg)/2 (1+xipg)/2];
Nxi_mef = [-1/2 1/2; -1/2 1/2];
% Total number of nodes and elements
numnp = size(xnode,2);
numel = numnp-1;
% Number of Gauss points on an element
ngaus = size(wpg,1);
% Allocate storage
A = zeros(numnp, numnp);
B = zeros(numnp,numnp);
f = zeros(numnp,1);
% MATRICES COMPUTATION
% Loop on elements
for i=1:numel
    unos = ones (ngaus,1);
    h = xnode(i+1)-xnode(i);
    xm = (xnode(i) +xnode(i+1))/2;
```

```
    weight = wpg*h/2;
    isp = [i i+1];
    % Loop on Gauss points (numerical quadrature)
    for ig = 1:ngaus
        N = N_mef(ig,:);
        Nx = Nxi_mef(ig,:)*2/h;
        w_ig = weight(ig);
        x = xm + h/2*xipg(ig); % x-coordinate of the current Gauss point
        % Matrices assembly
        A(isp,isp) = A(isp,isp) + w_ig*N'*N;
        B(isp,isp) = B(isp,isp) + w_ig*dt*(a*Nx)'*N;
        f(isp) = f(isp) + w_ig*dt_2*(N')*SourceTerm(x);
    end
```

end

## B Developed codes for 1D unsteady convection-diffusion

## B. 1 Modified function Galerkin.m

This function includes the implementation of Adams-Bashforth time integration scheme with Galerkin spatial formulation.

```
function Sol = Galerkin(T,s,a,nu,f,K,M,C,xnode,dt,nstep,c,Accd1,bccd1,d_temp)
% Sol = Galerkin(T,S,a,nu,f,K,M,G, xnode,dt,nstep, C,Accdl,bccdl)
% This function computes solution Sol at each time step using Galerkin ...
        formulation
%
% Input
% T,s: time-integration matrices
% tau: stabilization matrix
% a,nu: problem coefficients
% f,K,M,C: matrices obtained by discretizing the different terms ...
    Of the PDE using FEM
% xnode: vector of nodal coordinates
% dt: time step
% nstep: number of time steps to be computed
% C: initial condition
% Accdl, bccdl: matrices to impose boundary conditions using lagrange ...
    multipliers
% d_temp: Time discretization method (added for Adams-Bashforth ...
    scheme)
% Number of points
npoin = size(xnode,2);
if d_temp == 3 % Adams-Bashforth
    M_reduced = M(2:end-1, 2:end-1);
    Sol = c;
    for i=1:nstep
            if i == 1 % use forward Euler for 1st step
                    rhs = dt*(M*f - a*C*C - nu*K*C);
                dc = M_reduced\rhs (2:end-1);
                c(2:end-1) = c(2:end-1) + dc;
                Sol = [SOl c];
            else % use Adams-Bashforth for the rest of steps
                    rhs}=dt*(M*f-1.5*a*C*Sol(:, end) - 1.5*nu*K*Sol(:, end) + ...
                    0.5*a*C*Sol(:, end-1) + 0.5*nu*K*Sol(:, end-1));
                    dc = M_reduced\rhs (2:end-1);
                    c(2:end-1) = c(2:end-1) + dc;
                    Sol = [SOl c];
            end
        end
else
        % Integration matrix
        [n,m] = size(T);
        Id = eye(n,m);
        % Computation of the matrix necessary to obtain solution at each ...
            time-step: A du = F
        Kt =a*C + nu*K;
```

```
A = [];
    for i = 1:n
    row = [];
    for j = 1:m
        row = [row, Id(i,j)*M + dt*T(i,j)*Kt];
    end
    A = [A; row];
    end
    Mf = M*f;
    nccd = size(Accd1,1);
    Accd = []; bccd = [];
    for i = 1:n
    row = [];
    for j = 1:m
        row = [row, Id(i,j)*Accdl];
    end
    Accd = [Accd; row];
    bccd = [bccd; bccd1];
    end
    nccd = n*nccd;
    Atot = [A Accd'; Accd zeros(nccd)];
    % Factorization of matrix Atot
    [L,U] = lu(Atot);
    Sol = c;
    % Loop to compute the transient solution
    for i=1:nstep
        aux = dt*(-Kt*C + Mf);
        F = [];
        for i =1:n
        F = [F; S(i)*aux];
    end
    F = [F;bccd*0];
    dc = U\ (L\F);
    dc = reshape(dc(1:n*npoin),npoin,n);
    c = c + sum(dc,2);
    Sol = [Sol c];
    end
```

end

## References

[1] Donea, J., Huerta, A. (2004). Finite Element Methods for Flow Problems. Chichester: Wiley, pp. 141.
[2] Donea, J., Huerta, A. (2004). Finite Element Methods for Flow Problems. Chichester: Wiley, pp.128-129.
[3] Donea, J., Huerta, A. (2004). Finite Element Methods for Flow Problems. Chichester: Wiley, pp. 143.

