Assignment 4<br>Computational Structural Mechanics and Dynamics

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On "Isoparametric representation"

## 1 Assignment 4.1

A 3 -node straight bar element is defined by 3 nodes: 1,2 and 3 with axial coordinates $x_{1}$, $x_{2}$ and $x_{3}$ respectively as illustrated in figure below. The element has axial rigidity EA, and length $l=x_{1}-x_{2}$. The axial displacement is $\mathrm{u}(\mathrm{x})$. The 3 degrees of freedom are the axial node displacement $u_{1}, u_{2}$ and $u_{3}$. The isoparametric definition of the element is

$$
\left[\begin{array}{l}
1 \\
x \\
u
\end{array}\right]=\left[\begin{array}{ccc}
1 & 1 & 1 \\
x_{1} & x_{2} & x_{3} \\
u_{1} & u_{2} & u_{3}
\end{array}\right]\left[\begin{array}{l}
N_{1}^{e} \\
N_{2}^{e} \\
N_{3}^{e}
\end{array}\right]
$$

in which $N_{1}^{e}(\xi)$ are the shape functions of a three bar element. Node 3 lies between 1 and 2 but is not necessarily at the midpoint $\mathrm{x}=\mathrm{l} / 2$. For convenience define,

$$
x_{1}=0 \quad x_{2}=\left(\frac{1}{2}+\alpha\right) l \quad x_{3}=l
$$

where $-\frac{1}{2}<\alpha<\frac{1}{2}$ characterizes the location of node 3 with respect to the element center. If $\alpha=0$ node 3 is located at the midpoint between 1 and 2 .


Figure 1: The three-node bar element in its local system

1. Get the Jacobian $J=\frac{d x}{d \xi}$ in terms of $\mathrm{l}, \alpha$ and $\xi$. Show that,

- if $-\frac{1}{4}<\alpha<\frac{1}{4}$ then $J>0$ over the whole element $-1 \leq \xi \leq 1$.
- if $\alpha=0, J=\frac{l}{2}$ is a constant over the element.

2. Obtain the 1 x 3 strain displacement matrix B relating $e=\frac{d u}{d x}=B u^{e}$ where $u^{e}$ is the column 3 -vector of the node displacement $u_{1}, u_{2}$ and $u_{3}$. The entries of B are functions of $\mathrm{l}, \alpha$ and $\xi$.

### 1.1 Solution

For a quadratic Lagrange element with three nodes at $\xi_{1}=-1, \xi_{2}=0$ and $\xi=1$ the shape functions are:

$$
\begin{gathered}
N_{1}=\frac{1}{2} \xi(\xi-1) \\
N_{2}=\left(1-\xi^{2}\right) \\
N_{3}=\frac{1}{2} \xi(\xi+1)
\end{gathered}
$$

A parametric interpolation of the element geometry yields as follows:

$$
x=\sum_{i=1}^{3} N_{i}(\xi) x_{i}
$$

Substituting the shape functions and the values of $x_{i}$ :

$$
\begin{gathered}
x=\frac{1}{2} \xi(\xi-1)(0)+\left(1-\xi^{2}\right)\left(\frac{1}{2}+\alpha\right) l+\frac{1}{2} \xi(\xi+1) l \\
x=-\frac{l}{2}\left[2 \xi^{2} \alpha-\xi-2 \alpha-1\right]
\end{gathered}
$$

Where the Jacobian is defined as:

$$
J=\frac{d x}{d \xi}=-\frac{l}{2}[4 \xi \alpha-1]
$$

To show that $J>0$ when $-\frac{1}{4}<\alpha<\frac{1}{4}$ for the whole element $-1 \leq \xi \leq 1$, we have that:

$$
\begin{gathered}
|\alpha \cdot \xi|=|\alpha| \cdot|\xi|<\frac{|\xi|}{4} \leq \frac{1}{4} \\
|\alpha \cdot \xi|<\frac{1}{4} \\
\frac{-1}{4}<\alpha \cdot \xi<\frac{1}{4} \\
4 \alpha \cdot \xi<1 \\
4 \alpha \cdot-1<0 \\
l(4 \alpha \cdot-1)<0 \\
-l(4 \alpha \cdot-1)>0 \\
-\frac{l}{2}(4 \alpha \cdot-1)>0 \\
J>0
\end{gathered}
$$

If $\alpha=0$, J doesn't depend on $\xi$, therefore, it is a constant:

$$
\begin{gathered}
J=-\frac{l}{2}[4 \xi(0)-1]=-\frac{l}{2}[-1] \\
J=\frac{l}{2}
\end{gathered}
$$

The strain displacement matrix $B$ is defined as follows:

$$
B=\frac{d N}{d x}=J^{-1} \frac{d N}{d \xi}
$$

Where:

$$
\begin{aligned}
J^{-1} & =-\frac{2}{l(4 \xi \alpha-1)} \\
N & =\left[\begin{array}{lll}
N_{1} & N_{2} & N_{3}
\end{array}\right]
\end{aligned}
$$

$$
\begin{gathered}
\frac{N_{1}}{\xi}=\frac{2 \xi-1}{2} \\
\frac{N_{2}}{\xi}=-2 \xi \\
\frac{N_{3}}{\xi}=\frac{2 \xi+1}{2}
\end{gathered}
$$

Therefore:

$$
\left.\left.\begin{array}{c}
B=-\frac{2}{l(4 \xi \alpha-1)}\left[\begin{array}{lll}
\frac{2 \xi-1}{2} & -2 \xi & \frac{2 \xi+1}{2}
\end{array}\right] \\
B=\left[\begin{array}{llll}
-\frac{2 \xi-1}{l(4 \xi \alpha-1)} & \frac{4 \xi}{l(4 \xi \alpha-1)} & -\frac{2 \xi+1}{l(4 \xi \alpha-1)}
\end{array}\right] \\
B=\frac{1}{l(4 \xi \alpha-1)}[-(2 \xi-1)
\end{array}\right] \xi-(2 \xi+1)\right] . ~ \$
$$

On "Structures of revolution"

## 2 Assignment 4.2

1. Compute the entries of $K^{e}$ for the following axisymmetric triangle:

$$
r_{1}=0 \quad r_{2}=r_{3}=a, \quad z_{1}=z_{2}=0, \quad z_{3}=b
$$

The material is isotropic with $\nu=0$ for which the stress-strain matrix is,

$$
E=E\left[\begin{array}{cccc}
1 & 0 & 0 & 0 \\
0 & 1 & 0 & 0 \\
0 & 0 & 1 & 0 \\
0 & 0 & 0 & \frac{1}{2}
\end{array}\right]
$$

2. Show that the sum of the rows (and columns) 2,4 and 6 of $K^{e}$ must vanish and explain why. Show as well that the sum of rows (and columns) 1,3 and 5 does not vanish, and explain why.
3. Compute the consistent force vector $f^{e}$ for gravity forces $b=[0,-g]^{T}$.

### 2.1 Solution

The stiffness matrix can be computed according to the general relationship:

$$
K_{i j}^{e}=2 \pi \int B_{i}^{T} D B_{j} r d r d z
$$

Since the integration over the element is algebraically complex, the numerical integration fits better this case, evaluating all quantities for a centroidal point:

$$
\bar{r}=\frac{r_{i}+r_{j}+r_{m}}{3} \quad \text { and } \quad \bar{z}=\frac{z_{i}+z_{j}+z_{m}}{3}
$$

In this case the stiffness matrix approximation is defined as follows:

$$
K_{i j}^{e}=2 \pi \bar{B}_{i}^{T} D \bar{B}_{j} \bar{r} A
$$

Where the shape functions are and its derivatives are:

$$
\begin{array}{clll}
N_{1}=1-\frac{r}{a} ; & \frac{\partial N_{1}}{\partial r}=-\frac{1}{a} ; & \frac{\partial N_{1}}{\partial z}=0 \\
N_{2}=\frac{r}{a}-\frac{z}{b} ; & \frac{\partial N_{2}}{\partial r}=\frac{1}{a} ; & \frac{\partial N_{2}}{\partial z}=-\frac{1}{b} \\
N_{3}=\frac{z}{b} ; & \frac{\partial N_{3}}{\partial r}=0 ; & \frac{\partial N_{3}}{\partial z}=\frac{1}{b}
\end{array}
$$

Note: The values of " $r$ " and " $z$ " will have to be substituted by $\bar{r}$ and $\bar{z}$ respectively.
Computing the centroide of the element, we get:

$$
\bar{r}=\frac{2 a}{3} ; \quad \bar{z}=\frac{b}{3}
$$

The strain matrix is defined as follows:

$$
B_{i}=\left[\begin{array}{cc}
\frac{\partial N_{i}}{\partial r} & 0 \\
0 & \frac{\partial N_{i}}{\partial z} \\
\frac{N_{i}}{r} & 0 \\
\frac{\partial N_{i}}{\partial r} & \frac{\partial N_{i}}{\partial r}
\end{array}\right]
$$

Substituting, the following strain matrix is obtained:

$$
B=\left[\begin{array}{cccccc}
-1 / a & 0 & 1 / a & 0 & 0 & 0 \\
0 & 0 & 0 & -1 / b & 0 & 1 / b \\
1 /(2 * a) & 0 & 1 /(2 * a) & 0 & 1 /(2 * a) & 0 \\
0 & -1 / a & -1 / b & 1 / a & 1 / b & 0
\end{array}\right]
$$

Poisson ratio is considered to be 0 , therefore the constitutive matrix yields:

$$
D=E\left[\begin{array}{llll}
1 & 0 & 0 & 0 \\
0 & 1 & 0 & 0 \\
0 & 0 & 1 & 0 \\
0 & 0 & 0 & \frac{1}{2}
\end{array}\right]
$$

The area of the element is:

$$
A=\frac{1}{2} \operatorname{det}\left(\left[\begin{array}{lll}
1 & 0 & 0 \\
1 & a & 0 \\
1 & a & b
\end{array}\right]\right)=\frac{a b}{2}
$$

Now the stiffness matrix can be obtained:

$$
\begin{gathered}
K_{i j}^{e}=2 \pi \frac{2 a}{3} \frac{a b}{2} \bar{B}_{i}^{T} D \bar{B}_{j} \\
K_{i j}^{e}=\frac{2 \pi a^{2} b}{3} \bar{B}_{i}^{T} D \bar{B}_{j} \\
K_{i j}^{e}=\frac{2 \pi a^{2} b}{3} E\left[\begin{array}{cccccc}
\frac{5}{4 a^{2}} & 0 & \frac{-3}{4 a^{2}} & 0 & \frac{1}{4 a^{2}} & 0 \\
0 & \frac{1}{2 a^{2}} & \frac{1}{2 a b} & \frac{-1}{2 a^{2}} & \frac{-1}{2 a b} & 0 \\
\frac{-3}{4 a^{2}} & \frac{1}{2 a b} & \frac{5}{4 a^{2}}+\frac{1}{2 b^{2}} & \frac{-1}{2 a b} & \frac{1}{4 a^{2}}-\frac{1}{2 b^{2}} & 0 \\
0 & \frac{-1}{2 a^{2}} & \frac{-1}{2 a b} & \frac{1}{2 a^{2}}+\frac{1}{b^{2}} & \frac{1}{2 a b} & \frac{-1}{b^{2}} \\
\frac{1}{4 a^{2}} & \frac{-1}{2 a b} & \frac{1}{4 a^{2}}-\frac{1}{2 b^{2}} & \frac{1}{2 a b} & \frac{1}{4 a^{2}}+\frac{1}{2 b^{2}} & 0 \\
0 & 0 & 0 & \frac{-1}{b^{2}} & 0 & \frac{1}{b^{2}}
\end{array}\right]
\end{gathered}
$$

A first approximation if the body forces are constant is:

$$
\begin{gathered}
f_{i}^{e}=-2 \pi \mathbf{b} \frac{\bar{r} A}{3} \\
f_{i}^{e}=-2 \pi \mathbf{b} \frac{1}{3} \frac{2 a}{3} \frac{a b}{2}=-2 \pi \frac{a^{2} b}{9} \mathbf{b}
\end{gathered}
$$

Where $\mathbf{b}$, is:

$$
\mathbf{b}=\left[\begin{array}{c}
0 \\
-g
\end{array}\right]
$$

Therefore:

$$
f=2 \pi \frac{a^{2} b g}{9}\left[\begin{array}{l}
0 \\
1 \\
0 \\
1 \\
0 \\
1
\end{array}\right]
$$

## 3 Discussion

The stiffness matrix and consistent force vector was obtained through a centroidal approximation to avoid the complexity of the analytical integration, but it could have been done through analytical or numerical approximation as well.
Once the stiffness matrix is obtained, it can be seen that the sum of the rows (and columns) 2,4 and 6 vanishes, while the sum of the rows (and columns) 1,3 and 5 does not vanish. This is because in the " $z$ " direction there must be a balance for each axisimetry plane, while the balance in the " $r$ " axis direction will occur with the geometric counterpart of the problem. The gravity forces vector $\mathbf{b}$, has to be multiplied by the density of the material, therefore the consistent forces vector yields as follows:

$$
f=2 \pi \frac{a^{2} b \rho g}{9}\left[\begin{array}{l}
0 \\
1 \\
0 \\
1 \\
0 \\
1
\end{array}\right]
$$

