

Assignment 3

1)

$$1) \quad 1 + V = \frac{E}{2M}$$

$$V = \frac{E}{2M} - 1$$

$$\lambda = \frac{\left(\frac{E}{2M} - 1\right) E}{\left(\lambda + \frac{E}{2M} - 1\right) \left(1 - 2\left[\frac{E}{2M} - 1\right]\right)}$$

$$\lambda = \frac{\frac{E}{2M} (E - 2M)}{\frac{E}{2M} \left(1 - 2\left[\frac{E}{2M} - 1\right]\right)}$$

$$\lambda = \frac{E - 2M}{1 - \frac{2E}{2M} + 2} = \frac{(E - 2M) 2M}{2M - 2E + 4M} = \frac{M(E - 2M)}{3M - E}$$

$$3M\lambda - \lambda E = ME - 2M^2$$

$$E(\lambda + M) = M(3\lambda + 2M)$$

$$E = \frac{M(3\lambda + 2M)}{\lambda + M}$$

$$V = \frac{M(3\lambda + 2M)}{2M(\lambda + M)} - 1$$

$$V = \frac{3\lambda + 2M - 2\lambda - 2M}{2(\lambda + M)}$$

$$V = \frac{\lambda}{2(M + \lambda)}$$

2) Plane stress Matrix

$$E = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix}$$

Substituting E and ν , with λ and μ
we get:

$$E_{\lambda\mu} = 2\mu \begin{bmatrix} \frac{4(\mu+\lambda)}{2\mu+\lambda} & \frac{2\lambda}{2\mu+\lambda} & 0 \\ \frac{2\lambda}{2\mu+\lambda} & \frac{4(\mu+\lambda)}{2\mu+\lambda} & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

3) Plane ~~stress~~ strain Matrix

$$E = \frac{E}{(1+\nu)(1-2\nu)} \begin{bmatrix} 1-\nu & \nu & 0 \\ \nu & 1-\nu & 0 \\ 0 & 0 & \frac{(1-2\nu)}{2} \end{bmatrix}$$

substituting E and ν

$$E_{\lambda\mu} = \begin{bmatrix} 2\mu+\lambda & \lambda & 0 \\ \lambda & 2\mu+\lambda & 0 \\ 0 & 0 & \mu \end{bmatrix}$$

$$E_{\lambda\mu} = E_{\lambda} + E_{\mu}$$

$$E_{\lambda} = \begin{bmatrix} \lambda & \lambda & 0 \\ \lambda & \lambda & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

$$E_{\mu} = \begin{bmatrix} 2\mu & 0 & 0 \\ 0 & 2\mu & 0 \\ 0 & 0 & \mu \end{bmatrix}$$

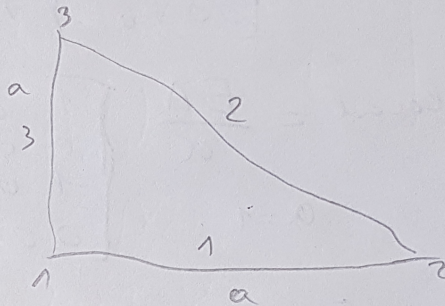
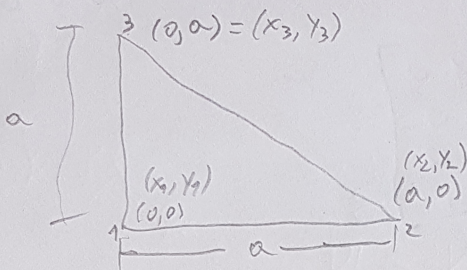
4) Express E_M and E_N in terms of E and ν

$$E_N(E, \nu) = \frac{E\nu}{(1+\nu)(1-2\nu)} \begin{bmatrix} 1 & 1 & 0 \\ 1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

$$E_M(E, \nu) = \frac{E}{(1+\nu)} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1/2 \end{bmatrix}$$

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$$a=1, h=1, \nu=0$$



$$B = DN = \frac{1}{2Ae} \begin{bmatrix} y_{23} & 0 & y_{31} & 0 & y_{12} & 0 \\ 0 & x_{32} & 0 & x_{13} & 0 & x_{21} \\ x_{32} & y_{32} & x_{13} & y_{31} & x_{21} & y_{12} \end{bmatrix} \begin{cases} y_{ij} = y_i - y_j \\ x_{ij} = x_i - x_j \end{cases}$$

$$Ae = \frac{1}{2} \det \begin{bmatrix} 1 & 1 & 1 \\ x_1 & x_2 & x_3 \\ y_1 & y_2 & y_3 \end{bmatrix}$$

$$E = \frac{E}{(1-\nu)^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix}$$

$$K_{tri}^e = \int_{\Omega} h B^T E B d\Omega = h B^T E B \int_0^{Ae} d\Omega = h B^T E B Ae$$

$$K_{tri}^e = h B^T E B Ae$$

$$k_{tri} = \begin{bmatrix} \frac{3E}{4} & \frac{E}{4} & -\frac{E}{2} & -\frac{E}{4} & -\frac{E}{4} & 0 \\ \frac{E}{4} & \frac{3E}{4} & 0 & -\frac{E}{4} & -\frac{E}{4} & -\frac{E}{2} \\ -\frac{E}{2} & 0 & \frac{E}{2} & 0 & 0 & 0 \\ -\frac{E}{4} & -\frac{E}{4} & 0 & \frac{E}{4} & \frac{E}{4} & 0 \\ -\frac{E}{4} & -\frac{E}{4} & 0 & \frac{E}{4} & \frac{E}{4} & 0 \\ 0 & -\frac{E}{2} & 0 & 0 & 0 & \frac{E}{2} \end{bmatrix} \leftarrow \begin{cases} a=1 \\ h=1 \\ \nu=0 \end{cases}$$

$$K_{1local} = \frac{EA}{a} \begin{bmatrix} 1 & 0 & -1 & 0 \\ 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}, \quad K_{2local} = \frac{\sqrt{2} EA}{2a} \begin{bmatrix} 1 & 0 & -1 & 0 \\ 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

$$K_{3local} = \frac{EA}{a} \begin{bmatrix} 1 & 0 & -1 & 0 \\ 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

$a=1$

$$K_1 = T_1^T k_{1local} T_1 = EA \begin{bmatrix} 1 & 0 & -1 & 0 \\ 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} \begin{matrix} u_{x1} \\ u_{y1} \\ u_{x2} \\ u_{y2} \end{matrix}$$

$$K_2 = T_2^T k_{2local} T_2 = \frac{\sqrt{2} EA}{4} \begin{bmatrix} 1 & -1 & -1 & 1 \\ -1 & 1 & 1 & -1 \\ -1 & 1 & 1 & -1 \\ 1 & -1 & -1 & 1 \end{bmatrix} \begin{matrix} u_{x2} \\ u_{y2} \\ u_{x3} \\ u_{y3} \end{matrix}$$

$$K_3 = T_3^T k_{3local} T_3 = EA \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 \\ 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix} \begin{matrix} u_{x3} \\ u_{y3} \\ u_{x4} \\ u_{y4} \end{matrix}$$

$$K_1 G = EA \begin{bmatrix} 1 & 0 & -1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \begin{matrix} u_{x1} \\ u_{y1} \\ u_{x2} \\ u_{y2} \\ u_{x3} \\ u_{y3} \end{matrix}$$

$$K_2 \epsilon = \frac{EA\sqrt{2}}{4} \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & -1 & 1 & -1 \\ 0 & 0 & -1 & 1 & 1 & -1 \\ 0 & 0 & -1 & 1 & 1 & -1 \\ 0 & 0 & 1 & -1 & 1 & -1 \end{bmatrix} \begin{matrix} u_{x1} \\ u_{y1} \\ u_{x2} \\ u_{y2} \\ u_{x3} \\ u_{y3} \end{matrix}$$

$$K_3 \epsilon = EA \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & -1 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 0 & 0 & 1 \end{bmatrix}$$

$$K_{bar} = K_1 \epsilon + K_2 \epsilon + K_3 \epsilon = k_{i\alpha} \epsilon \in \{1, 2, 3\}$$

$$K_{bar} = \begin{bmatrix} EA & 0 & -EA & 0 & 0 & 0 \\ 0 & EA & 0 & 0 & 0 & -EA \\ -EA & 0 & \frac{EA(\sqrt{2}+4)}{4} & -\frac{EA\sqrt{2}}{4} & -\frac{\sqrt{2}EA}{4} & \frac{\sqrt{2}EA}{4} \\ 0 & 0 & -\frac{\sqrt{2}EA}{4} & \frac{\sqrt{2}EA}{4} & \frac{\sqrt{2}EA}{4} & -\frac{\sqrt{2}EA}{4} \\ 0 & 0 & -\frac{\sqrt{2}EA}{4} & \frac{\sqrt{2}EA}{4} & \frac{\sqrt{2}EA}{4} & -\frac{\sqrt{2}EA}{4} \\ 0 & -EA & \frac{\sqrt{2}EA}{4} & -\frac{\sqrt{2}EA}{4} & -\frac{\sqrt{2}EA}{4} & \frac{EA(\sqrt{2}+4)}{4} \end{bmatrix}$$

2) N_0 ; I_t $A_1 = A_3 = \frac{3}{4}$ $A_2 = \frac{\sqrt{2}}{2}$

3) The triangular matrix has more non-zero terms, meaning that the DOF are more intricate than the truss bar global matrix. In the triangular stiffness matrix almost all elements depends on at first line contributes.

1) $\nu \neq 0$

The bar, truss is not affected by this proposition.

The increase of ν to 1 make several position of the matrix $\rightarrow \infty$.

$$\begin{bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{xy} \end{bmatrix} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \begin{bmatrix} \epsilon_{xx} \\ \epsilon_{yy} \\ 2\epsilon_{xy} \end{bmatrix}$$

$$\sigma_{xx} = \frac{E}{1-\nu^2} [\epsilon_{xx} + \nu \epsilon_{yy}] \rightarrow \nu=0 \rightarrow \sigma_{xx} = E \epsilon_{xx}$$

But $\nu \neq 0$ the longitudinal component (ϵ_{xx}) play a bigger role in composition to $\nu=0$ and the transverse component starts to get important. Let's say $\nu=0.2$

$$\sigma_{xx_2} = \frac{E}{1-0.04} [\epsilon_{xx} + 0.2 \epsilon_{yy}]$$

$$\sigma_{xx_2} = E [1.042 \epsilon_{xx} + 0.208 \epsilon_{yy}]$$

For the same strains, stresses with higher poisson ratio are also higher.

For same stresses the strains are smaller.
If $\nu > 1$ several terms of the matrix diagonal becomes negative.